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A CAVITY RADIOMETER FOR EARTH ALBEDO MEASUREMENT

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TECHNICAL MEASUREMENTS, INC.
La Canada, California

EXECUTIVE SUMMARY

A CAVITY RADIOMETER FOR EARTH ALBEDO MEASUREMENTS

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Radiometric measurements of the directional albedo of the Earth requires a detector with a flat response from 0.2 to 50 microns, a response time of about two seconds, a sensitivity of the order of 0.02 mw/cm^2 , and a measurement uncertainty of less than 5%. Absolute cavity radiometers easily meet the spectral response and accuracy requirements for Earth albedo measurements, but the radiometers available today lack the necessary sensitivity and response time.

In this effort, the specific innovations addressed were the development of a very low thermal mass cavity and printed/deposited thermocouple sensing elements which were incorporated into the radiometer design to produce a sensitive, fast response, absolute radiometer. The new cavity is applicable to the measurement of the reflected and radiated fluxes from the earth's surface and lower atmosphere from low earth orbit satellites. The effort consisted of requirements and thermal analysis; design, construction, and test of prototype elements of the black cavity and sensor elements to show proof-of-concept.

The results obtained indicate that a black body cavity sensor that has inherently a flat response from 0.2 to 50 microns can be produced which has a sensitivity of at least 0.02 mw/cm^2 per micro volt output and with a time constant of less than two seconds. Additional work is required to develop the required thermopile.

TECHNICAL MEASUREMENTS, INC
La Canada, California

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PROJECT SUMMARY

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In this effort, the specific innovations addressed were the development of a very low thermal mass cavity and printed/deposited thermocouple sensing elements which were incorporated into the radiometer design to produce a sensitive, fast response, absolute radiometer. The new cavity is applicable to the measurement of the reflected and radiated fluxes from the earth's surface and lower atmosphere from low earth orbit satellites. The effort consisted of requirements and thermal analysis; design, construction, and test of prototype elements of the black cavity and sensor elements to show proof-of-concept.

The results obtained show that with further development, a black body cavity sensor can be produced which meets all of the critical requirements for a top-of-the-atmosphere earth radiation budget instrument.

PROJECT OBJECTIVE

I. General

The end objective of the multiphased SBIR effort is the development of an Absolute Cavity Array Instrument (ACAI) suitable for measuring the top of the atmosphere flux from the full earth disk visible from an orbiting satellite at approximately 830km altitude. The ACAI would be composed of an number of detectors mounted in an array to measure the radiation field below the satellite. By using an array of fixed sensors of the absolute cavity type,

problems associated with mechanical motion and spectral response are eliminated. The advantages to be derived from the use of the absolute cavity sensors should be the following:

1. Measurements needing no spectral corrections.
2. Self-calibrating, direct energy measurements, with high stability.
3. Measurements directly comparable to ERB measurements.
4. A mechanically simple device, without moving mirrors.

The current phase of the research is centered on the cavity sensor: the feasibility of providing an absolute cavity sensor with the required sensitivity and response time to make the required measurements. The next phase of the research will address the integration of the sensor cavity into an instrument capable of making the desired measurements of earth TOA albedo. And then, the research will address issues associated with inflight calibration, field-of-view and spatial convolution, and overall instrument measurement accuracy and stability.

II. Sensor Requirements

For the ACAI to have the desired sensitivity, the cavity sensor must provide an electrical signal greater than 1 microvolt for the unit of energy corresponding to the required resolution. This equates to a sensitivity of at least $0.02 \text{ mw/cm}^2/\mu\text{v}$. The standard Kendall cavity used in the Mk VI radiometer has a sensitivity of only $.06 \text{ mw/cm}^2/\mu\text{v}$. Therefore an improvement of over three is required and a target improvement of 10 is desired.

The stated requirement for response time ($1/e$) is less than two seconds. The standard Kendall cavity exhibits a response time of approximately 6 seconds. Again, an improvement of at least three is required. This requirement is the most difficult to meet.

The spectral sensitivity of the cavity is required to be flat from .25 to 50 microns. The inside surface of TMI cavities are coated with 3M black velvet. This coating is not specular, and has a certain amount of retro-reflectance.

Measurements of emittance of this type cavity were made by NBS on a Mk VI cavity using a laser for irradiance and a conical pyrolytic receptor to measure the emittance. The results of this test gave an overall reflectance of 0.0021. This corresponds with analytic studies performed by JPL and TMI. A correction factor is applied to account for this loss of radiant power. No direct measurements of spectral sensitivity have been made, but it is quite certain that the spectral response is flat over the solar spectrum. The radiometer has been tested by the radiance from a black body warmed to a temperature of 10°C above the temperature of the radiometer (at 30°). The radiometer gives a very strong response, indicating spectral sensitivity at long wavelengths.

III. Optimum Cavity Sensor

The optimum cavity design for the ACAI represents a trade of size, sensitivity, and time response. It is beyond the scope of this effort to optimize a cavity design for ACAI, but it is within scope to determine the controlling parameters, to verify them, and to demonstrate the feasibility of achieving a satisfactory sensor. The reason that an optimum design can not be obtained at this stage is that the performance of the sensor is heavily dependent on the design of the view limiter and of the containing body. These, in turn, are driven by considerations of weight, size, and experiment design. In this effort we intend to show that we have sufficient control of the important variables and that instruments which can meet ACAI requirements can be produced.

WORK PERFORMED

I. Requirements Analysis

The scope of the requirements analysis is to establish an acceptable approach to the evaluation of the effect of instrument time constant on the interpretation of the data obtained from an open aperture cavity radiometer swept across a variable scene. This, in a reverse application, can provide the specification of time constant (together with other instrument parameters)

required for the interpretation of gathered experimental data with respect to the achievement of stated scientific objectives.

In a "shuttered" application where the "scene" goes from one fixed, constant value to a second fixed value, the conventional time constant is a direct measure of the required settling time to reach the second value within a given error of the measurement. For a continuously changing scene or a swept scene, a meaningful measure of the time-related accuracy is different and we suggest consideration of the steady state error in following a step change in the scene during the time the step propagates through the scene. This concept will now be developed.

A one dimensional model is assumed. For given values of radiometer aperture width, view limiter width, distance to view limiter and finally operating altitude, two scene or object sizes are determined. The first and smaller, width s_1 , is that region which is seen by the entire measurement aperture of the instrument. The second, total width s_2 to outer edges, are the additional regions on both sides of the first; these do not fill the measurement aperture because they are cut off in various amounts by the view limiter. Now define t_1 and t_2 as s_1/v and s_2/v where v is the (orbital) velocity of the sweeping motion.

Next consider the effect of the step change in radiated power as it propagates through the instrument view due to orbital motion. The step is perpendicular to the direction of motion. As the instrument is swept across this location, the region s_1 provides a ramp input of time length t_1 . The instrument also sees a pre-ramp and a post-ramp with a smaller slope; the duration is t_2 , which is symmetrically spaced around t_1 . The actual response of the instrument will be between that of the response to a single, unit amplitude, ramp of duration t_1 , and to a single, unit amplitude, ramp of duration t_2 greater than t_1 .

The response of a single time constant device (time constant = t_0) to a ramp input of unspecified duration of the shape or magnitude $W_0 t/t_1$ is:

$$W(t) = (W_0/t_1) \{ t - t_0 + t_0 \exp(-t/t_0) \}$$

It is readily seen that $W(0) = 0$, as it must be. Here $t = 0$ is the time when the ramp starts, i.e. when the edge of the ramp comes into view. For large t , t larger than t_0 , the instrument output eventually follows the ramp input with a simple time delay equal to t_0 , the instrument time constant.

For the above to be applicable to the experimental conditions here where the ramp times (t_1 or t_2) are finite, the value of t_0 must be significantly smaller than the ramp time; the value of t_0 must be at least less than $0.2 t_1$ and preferably less than $0.1 t_1$. (For the case of t_0 larger than t_1 the instrument is so slow that it can not tell the difference between a ramp input and a "shuttered" step input.)

Thus, for the case here of following with time delay t_0 , the error, w , in following the actual power input to the instrument (which is all it can be expected to measure) from a step change of power, W , at the scene is given by:

$$w / W = t_0 / t_1$$

This error exists during the center portion of the ramp. The transient departure at the start and the transient closure at the end of the ramp are of the form $\{1 - \exp(-t/t_0)\}$. For $t_2 = m t_1$, the error may be considered bounded: not as much as t_0/t_1 but not less than $t_0/t_2 = t_0 / m t_1$.

We now advocate the incorporation of the above description of instrument time related error to a step change at the scene into the combined tasks of the specification and design of, 1) the instrument to make the measurement, 2) the conditions under which the experiment is to be carried out, and 3) the interpretation of the acquired data.

A point on the curve is now given. With a scene spot size of 250 km viewed from an altitude of 830 km where the orbital velocity is about 7.4 km/sec:

$$t_1 = 250/7.4 = 33.8 \text{ sec}$$

For a step change of $1/2$ full scale in the scene (clouds. no clouds during sunlight) and an instrument time constant of 3.38 sec the error in not following the step is:

$$(1/2) (3.38/33.8) = 1/20 = 5\% \text{ of full scale}$$

The above analysis has focused entirely on the effect of instrument time constant. It must be remembered that the time constant "tracking" error, just discussed, does not relate or cause an error of the integral of the energy of the step change of radiation, but only an additional smearing of the step; the finite width of the aperture with its view limiter wings causes the first smear. Additionally, the amount of motion smear and instantaneous measurement error can be estimated from the rate-of-change in the recorded data for a clean, abrupt, single step change. The absolute sensitivity and resolution of the instrument is certainly of great importance; the physical reality is that these latter are related directly to the size of the instrument whereas the time constant increases (undesirably so) as the square of the linear size. The intent of this requirements analysis is to provide a useable measure of the effect of time constant for combination with other instrument and experiment requirements.

II. Thermal Analysis

A major effort was made to produce a theoretical, mathematically rigorous description of the transient thermal flow through simplified geometries that could adequately represent actual cavity and thermal resistor configurations. It was intended that this be used in meaningful evaluations of the model cavities to be produced using thin wall electroforming techniques.

The basic coordinate system for the constant wall thickness, conical receptor is cylindrical: r , θ and z . (Cut out a pie shaped portion of a thin flat pancake and form the balance into a cone.) The only spatial variable is the radial distance r ; flow is along lines of constant θ , and the walls are sufficiently thin, z direction, that is assumed there is negligible temperature difference and flow in that direction. The flow along the r direction, of course, is not spatially uniform; it diverges or converges with radial distance.

To start, the basic single flow parameter equations for the three systems were

set down: linear (Cartesian), cylindrical, and spherical (for variable thickness along cavity walls). The form of the basic time constant was determined for each system (referenced later in this report). Most of this effort was the gathering together of classical mathematical solutions available in many places; the purpose was to assemble the tools that could be useful in analysis of actual cavity and receptor configurations.

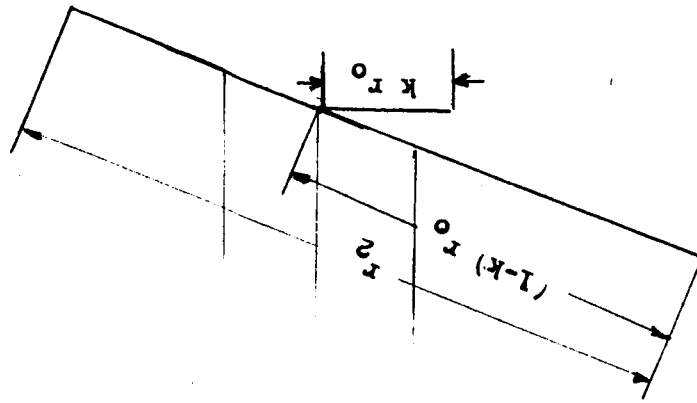
During this time, the concept emerged of a simple cone receptor with a mid-point thermal resistor as the minimum time constant configuration. One of the major driving forces was the fact that it was felt that this geometry could be quite rigorously analyzed. The measured performance should compare well with the prediction for thermal time constant. The balance of the thermal analysis performed here is based on this configuration, using the Bessel function solutions to the cylindrical coordinate configuration.

The mid-point thermal resistor configuration is built up from two conical pieces. The outer portion of the inner cone, which contains the closed apex, is folded over and down to form (a part of) the thermal resistor. The outer cone has a hole as its inner edge, and its lower portion is slit and split outward to form (the balance of) the thermal resistor. In the combined parts of the resistor the excess material from the inner cone fills the vacant spaces between the slits in the outer cone; the result is a cylindrical thermal resistor of constant thickness equal to twice that of conical, receptor, portion. The details are given in Figure 1. The rigorous transient thermal analysis model consists of starting at time zero with a representative temperature profile along the conical sections. This profile tapers to a constant reference temperature at the lower edge of the thermal resistor; the local temperature is measured as a temperature difference from this reference. At time equals zero the heat flux producing the initial temperature profile is considered to be removed and the temperature falls to the constant reference temperature at a rate indicative of the thermal time constant.

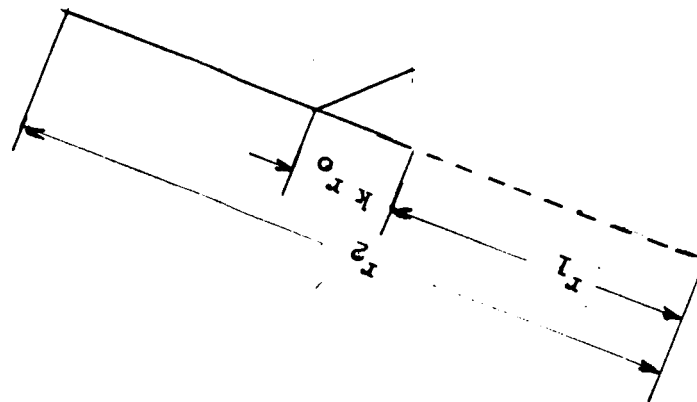
The mathematical form for the cylindrical case is well known and the general solution is given by:

$$T(r,t) = \sum \exp(-\alpha^2 \beta_s^2 t) \{A_s J_0(\beta_s r) + B_s Y_0(\beta_s r)\}$$

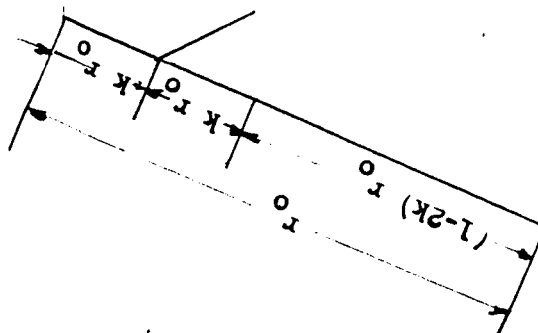
COMBINED
CONES



OUTER CONE



INNER CONE



DEVELOPMENT OF TWO CONE GEOMETRY

FIGURE -1

where: $\alpha^2 = k/cp =$ conductivity/volume heat capacity
 $J_0, Y_0 =$ Bessel functions of zero order
 $\beta_s =$ series of inverse lengths required to satisfy the reference temperature condition
 $A_s, B_s =$ series of temperatures magnitudes that satisfy the initial temperature distribution:

$$T(r,0) = \sum A_s J_0(\beta_s r) + B_s Y_0(\beta_s r)$$

When applied to the inner cone with a total length r_0 including the thermal resistor portion and containing the center apex at $r = 0$, first all B_s must be zero since $Y_0(0)$ is infinite. Next since, $T(r_0, t) = 0$ it is required that $J_0(\beta_s r_0) = 0$ and thus $(\beta_s r_0) = j_{0,s}$ where $j_{0,s}$ are the zeros of J_0 . Defining $t_s = 1/\alpha^2 \beta_s^2$ the equation for T becomes:

$$T(r,t) = \sum A_s e^{-t/t_s} J_0(j_{0,s} r/r_0)$$

where $t_s = r_0^2 / (j_{0,s})^2 =$ time constants of the components.

For the several components of the solution a short table of relative time constants demonstrates the concept of the dominant time constant which will be used subsequently:

s	$j_{0,s}$	t_s/t_1
1	2.4048	1.000
2	5.5201	.190
3	8.6537	.077
4	11.7915	.0415

Thus it is seen that the component terms for $s = 2, 3, 4$, etc. decay much more rapidly than for $s = 1$ and the t_1 time constant will dominate the transient behavior for the temperature profiles to be encountered in real cavities. This is discussed further shortly.

It will be illustrative to consider briefly other coordinant systems, Cartesian (linear) and spherical. The linear case represents a two dimensional trough shaped cavity and the spherical the case of cavity thickness zero at center apex and increasing linearly with distance outward.

For the linear case with a length of x_0 to the center of the cavity:

$$T(x,t) = \sum A_n \exp(-\alpha^2 t n^2 \pi^2 / 2^2 x_0^2) \sin(n\pi x/2x_0)$$

The n here clearly represents a series of harmonics, and for $dT/dx = \text{zero}$ at $x = x_0$ at the center of the "trough", n must be odd. The time constants are:

$$t_n = (x_0 / \alpha^2) / (n\pi/2)^2$$

For the spherical geometry case the solutions are of the form:

$$\sin(n\pi x/x_0)/(x/x_0).$$

The zeros are at $nx = \pi$ and the resulting time constants are:

$$t_n = (x_0^2 / \alpha^2) / (n\pi)^2$$

A tabular comparison of the dominant time constants shows the effect of geometry.

Geometry	Linear	Cylindrical	Spherical
Math form	Trig	Bessel	$(\sin x)/x$
Dominant time constant	$(2/\pi)^2$ $= 4/\pi^2$	$(1/j_{0,1})^2$ $= 1.708/\pi^2$	$(1/\pi)^2$ $= 1/\pi^2$

For the cylindrical case the dominant time constant is less than half that of the trough and for the spherical case it would be only one fourth. The spherical case, with its tapered walls, was not considered for implementation at this time.

The linear geometry case can be used to show quite easily that the dominant time constant is indeed dominant for representative initial temperature distributions. The initial temperature distribution is given by:

$$T(x,0) = \sum A_n \sin(n\pi x/2x_0) , n \text{ odd only}$$

By combining small amounts of 3rd, 5th, etc. harmonics, the temperature rise can be made quite linear from $x = 0$ to some fraction of x_0 , e.g. $1/3$. This corresponds to the thermal resistor portion where the constant heat flux produces a linear temperature drop. Beyond this "resistor" region the temperature profile continues to rise to a maximum at x_0 , the center of the trough. This is precisely the case for some form of sensibly constant heat flux absorbtion in this, the receptor, portions of the device. The relative time constants decrease as $1/9$, $1/25$, $1/49$, etc. of the dominant time constant. Thus even if there are significant quantities of the harmonics (relative to the $n=1$ fundamental), these will die out rapidly. The ultimately dominant transient temperature distribution will be the quarter period sinusoid that will decay at the dominant, $n=1$, time constant rate. The Bessel functions are basically trigonometric functions whose amplitude falls off with distance (approach $\sin x / x$ for large x); the spherical case behaves as $\sin x/x$. The concept of harmonics and the behavior of these harmonics is directly transferable from the linear, simple trigonometric, case to these.

The basic analytical work presented in the several previous paragraphs for the inner cone can be found in many text books. The concept of the dominant time constant and the representative initial temperature distributions has been specifically tailored to the circumstances of cavity receptors. The analysis to follow for the outer cone portion is considered to be new work, original to the specific requirements here.

For the outer cone, the task is to determine the dominant time constant for a two parameter system defined by an inner edge radial distance r_1 and an outer edge radial distance, r_2 instead of the single parameter r_0 for the inner cone. The boundary condition at r_1 is constant (reference) temperature since this edge will ultimately be the bottom of the thermal resistor. At r_2 , the outer edge of this outer cone, the requirement is that $dT/dr = 0$; no heat is conducted along the cone at this point since this is an edge that is thermally insulated. The Y_0 Bessel solution is now acceptable since r is always larger than zero. The single solution for the dominant time constant will be sought rather than the general sum of all the possible "harmonics". The general solution for the outer cone is:

$$T(r,t) = \exp(-t/t_0) \{ A_0 J_0(\beta_0 r) + B_0 Y_0(\beta_0 r) \}$$

First choose A_0, B_0 and T_0 to give the form:

$$T(r,t) = T_0 \exp(-t/t_0) \{ Y_0(\beta_0 r_1) J_0(\beta_0 r) - J_0(\beta_0 r_1) Y_0(\beta_0 r) \}$$

This satisfies the requirement $T(r_1, t) = 0$. The value of β_0 now must be found so that the value of the function at r_2 is the first maximum after the positive zero crossing at r_1 . The sign of T_0 is selected to give a positive going zero crossing. This value of β_0 satisfies the boundary condition at r_2 , the outer edge. The computational technique is as follows: (There are many excellent tables of J_0 and Y_0 so that the computational effort is quite minimal.)

- 1) Select a numerical value $x_1 = \beta_0 r_1$
- 2) Look up $J_0(x_1)$ and $Y_0(x_1)$
- 3) Locate $x_2 = \beta_0 r_2$ such that the function is a maximum at x_2 .

This value x_2 is a "quarter period" beyond x_1 based on the trigonometric analogy and a precise value could be determined with a very few iterations. The value of $x_2/x_1 = r_2/r_1$ defines the geometry of the outer cone. The time constant is now give by:

$$t_0 = 1/\alpha^2 \beta_0^2 = r_1^2 / \alpha^2 (\beta_0 r_1)^2 = r_1^2 / \alpha^2 x_1^2$$

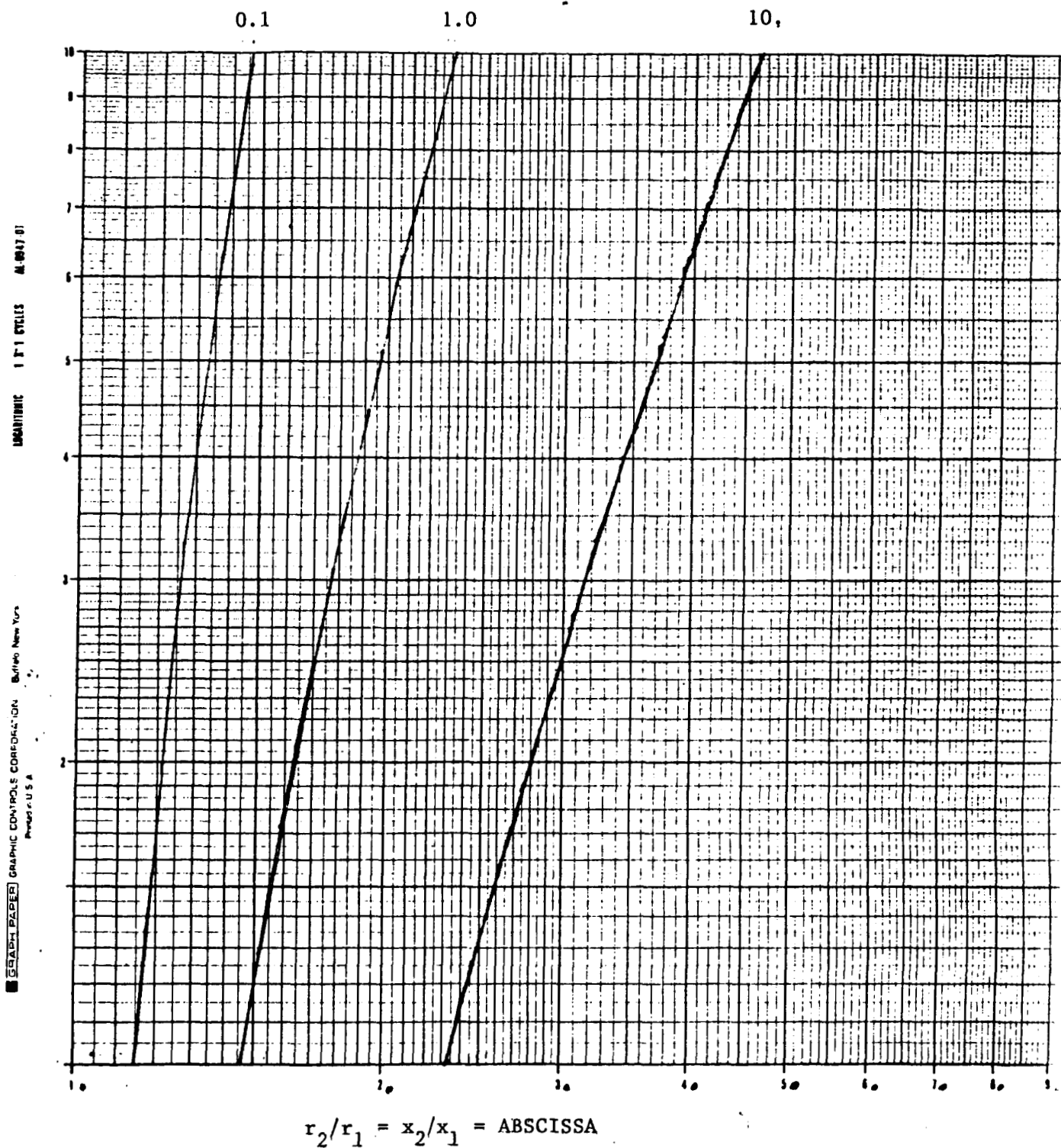
This $1/x_1^2$ defines the non-dimensional time constant Figure 2. is a plot of $1/x_1^2$ versus $x_2/x_1 = r_2/r_1$. This curve is considered to be one of the significant outputs of this analysis. For use in the next analysis steps the curve defines the functional relationship, F:

$$1/x_1^2 = F(x_2/x_1) = F(r_2/r_1) = F$$

The task now is to put together two cones, an inner and an outer, with equal dominate time constants such that a continuous smooth inside conical receptor surface is formed and the folded portions combine to form a cylindrical thermal resistor. The temperature drop across the thermal resistor is measured with a thermopile. The thermal resistor will be twice the thickness of the conical receptor surface. The length of the inner cone is r_0 and is

TOP OF SCALE VALUE FOR $F = 1/x_1^2 = \text{ORDINATE}$

ORIGINAL PAGE IS
OF POOR QUALITY



OUTER CONE DIMENSIONLESS TIME CONSTANT

FIGURE -2

the basic reference dimension. A length $k x_o$ of this cone is folded over to produce a thermal resistor of length $k x_o$. This requires that $r_1 = (1 - 2k)r_o$ so that when an equal length, $k r_o$, of the outer cone is folded out to complete the thermal resistor, the two cones match to form a smooth receptor surface. The inner and outer time constants which are set equal are given by:

$$t_{in} = r_o^2 / \alpha^2 (j_{o,1})^2 = 0.1729 r_o^2 / \alpha^2$$

$$t_{out} = r_1^2 / \alpha^2 x_1^2 = F r_1^2 / \alpha^2$$

$$F = 0.1729 (r_o/r_1)^2 = 0.1729 / (1 - 2k)^2$$

The combined cone has now been defined by two parameters, r_o and k . Small values of k will give large values of r_2 , the "aperture size" of the cone, with resulting large energy input but short length thermal resistors with small temperature drops for measurement by the thermopile. Smaller values of k decrease the aperture area and energy received but increase the thermal resistor length and the output voltage. In the next paragraphs the maximum (relative) sensitivity is found.

In high absolute cavity radiometers, a deep cavity is used. This provides a high cavity enhancement factor the reduces the effect of lack of precise knowledge of the emissivity of the receptor surface. However, the larger size and longer heat flow paths greatly increase the time constant. For here a simple conical shape is proposed with half cone angles of 30 degrees to provide at least some cavity enhancement. The following sensitivity analysis is based on a constant cone angle and hence produces a relative sensitivity for the constant half angle, ϕ , as a function of the length of the thermal resistor and the time constant. At a given flux density input:

$$\text{Power} = P_o (r_2 \sin \phi)^2 = P_o \sin^2 \phi (r_2/r_o)^2 r_o^2$$

$$\text{Circumference of thermal resistor} = 2 (1 - k)r_o \sin \phi$$

$$\text{Length of thermal resistor} = k r_o$$

Temperature drop across thermal resistor is proportional to:

$$(\text{Power})(\text{Length})/(\text{Circumference})$$

Number of thermocouples is proportional to the circumference.

Voltage produced is thus proportional to:

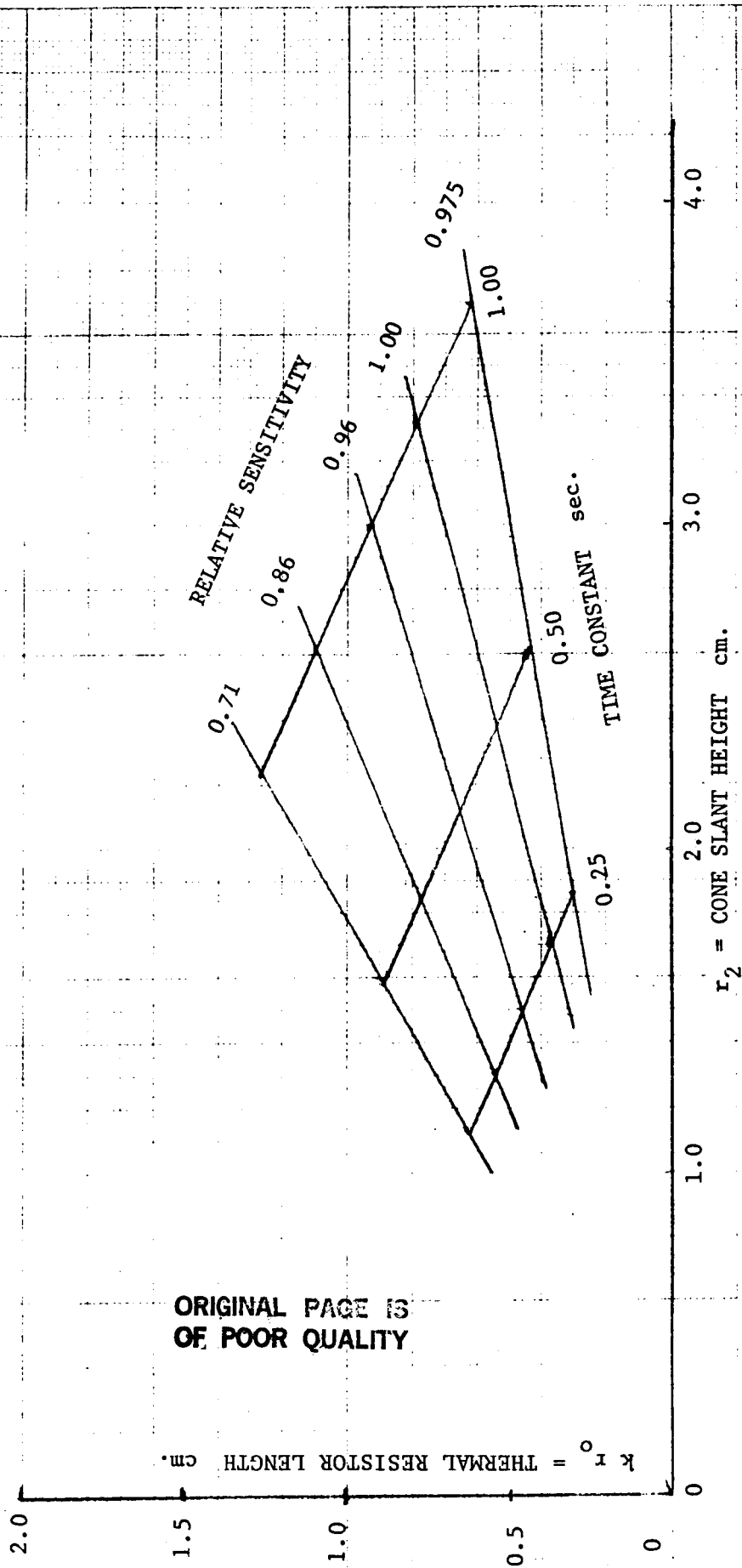
$$\begin{aligned} (\text{Temp drop})(\text{Circumference}) &= (\text{Power})(\text{Length}) \\ &= \sin^2 \phi \, r_2/r_o)^2 \, r_o^2 \, (k \, r_o) \\ &= \sin^2 \phi \, r_o^3 \, \{ k \, (r_2/r_o)^2 \} \end{aligned}$$

Thus at constant half cone angle and at a given time constant (which determines the value of r_o) the maximum electrical output is at the maximum of the product $k \, (r_2/r_o)^2$. This product maximizes at $k=0.252$. Silver is now selected as the cone and resistor material with its low value of α :

$$1/\alpha^2 = 0.580 \text{ sec} / \text{cm}^2$$

The graph of Figure 3 is now produced from these relationships for time constants of 0.25, 0.50 and 1.0 seconds. For half cone angles of 30° The slant length of the cone is also equal to the diameter at the outer edge. A long thermal resistor is desired for several reasons, e.g. attachment of the required electrical insulation barrier to isolate the thermopile junctions and attachment of the thermopile itself. The graph shows that the resistor length can be sensibly increased (also a smaller aperture and thus smaller instrument size itself is obtained) with only a modest and probably a very acceptable decrease in relative sensitivity. This curve is considered to be the second significant output of this analysis.

In summary, the basic objective of this transient thermal analysis was to provide tools and guidance for the selection of the size and geometry of the ultra light receptor cones and associated mid-point thermal resistors. Both the magnitude of the time constant and the geometric features for minimizing this magnitude were obtained. The term "mid-point" is a misnomer; in actual practice the location of the resistor is well toward the outer edge of the cone; thermal flux flow along the diverging inner portion is less impeded than the converging flow from the outer portion. Questions and concerns about the proper matching of the initial temperature conditions in the combining of the two folded parts to make the thermal resistor are valid. However, it is felt that the dominant time constant concept is a valid approach for the purpose here; "higher harmonic" terms do exist in an actual initial temperature distribution caused by the actual input thermal flux. But these will decay rapidly. This dominant time constant concept is intrinsic to this analysis



MID-POINT RECEPTOR GEOMETRY

FIGURE -3

and is considered a third significant analysis output.

III. Cavity Design

Modification of a standard black cavity sensor both by changes in geometry and by new technological advances in fabrication methods were explored. Two cavity designs were considered: the standard TMI design modified only for manufacturing by new methods, and a new cone cavity design attempting to achieve maximum performance with sensitivity greatly increased and time constant significantly decreased. Changing cavity shape is a trade of response time for cavity absorptivity, and uncertainty of measurement in long term use. The standard Kendall cavity design has an enhancement of a factor of six over a flat plate and a factor of 2 over a 30° cone. An additional factor that can be considered is giving up the self-calibration feature to gain response time.

A study of cavity geometry considering parameters of aperture area, wall thickness, (thermal mass) of the flux receptor versus the thermal resistance and diffusivity of metals as well as the geometry of the thermal resistor was made. The geometry of the combination of thermal resistor and flux receptor was also considered with respect to optimization of time constant and transient effects. The resulting design was a cavity sensor configuration consisting of two symmetrical cone-thermal resistor sensors mounted back to back on a heat sink temperature disc (See Figure 4). The function of this disc is to provide a reference temperature to which the temperature gradient caused by incoming irradiance can be measured. The opposite or back looking sensor is covered so that it sees no incoming irradiance but is used to compensate for any heat flux which may be entering or leaving the common mounting disc. The electrical signal from the "compensating" cavity sensor is connected so as to oppose the signal from the measuring cavity. If the two back-to-back cavity sensors are perfectly physically matched any signals caused in the two sensors by heat flux entering or leaving the heat sink mounting disk will be cancelled and not show up in the measurement.

The manufacture of perfectly symmetrical cavity and thermal resistor sensors is made feasible by using the electroforming process for fabrication. This

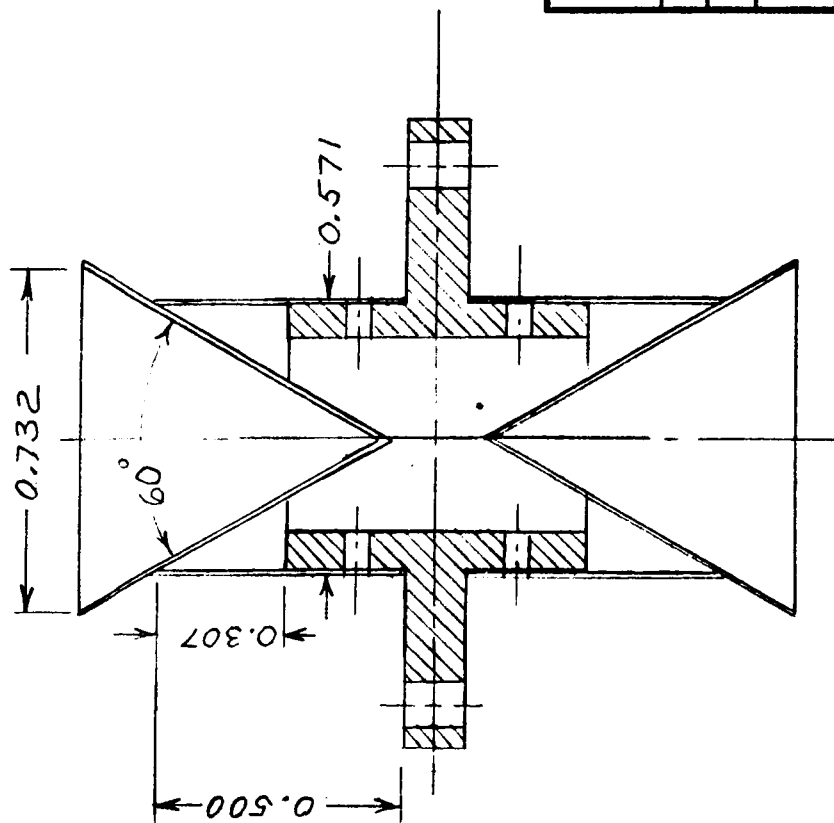
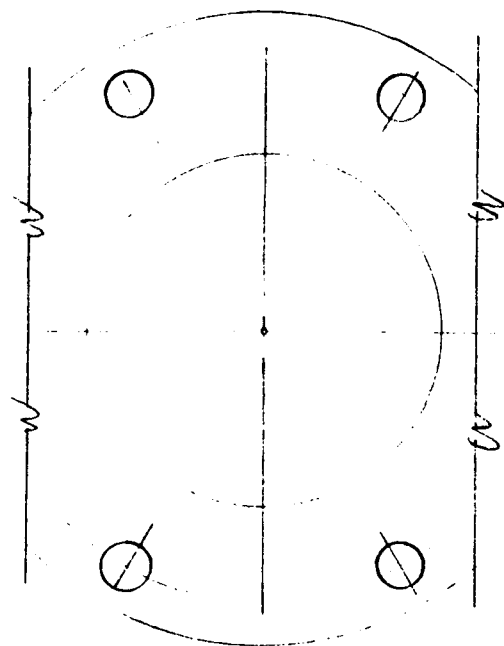


FIGURE 4

MATERIALS: CONES: 0.005 FINE SILVER
CYL. RESISTOR: 0.005 STERLING
MOUNT: COPPER
THERMOPILES: TYPE 001, 2 REQ'D, CHR-CON
APPROX. 160 JCT PAIR
SOLDERS: CONE TO CYL - 95-5 (430°F), LAST
JOINTS WITH MOUNT HEATSINKED
CYL. RES. TO MOUNT, FIRST JOINT
WITH 63-37

ORIGINAL PAGE IS
OF POOR QUALITY

TECHNICAL MEASUREMENTS, INC.
LA CANADA, CA

APPROVED BY:

SCALE: X 4

DATE: 6-1-87

DRAWN BY M.B.

REVISED

HIGH SENSITIVITY FAST TIME CONSTANT
CONE RADIOMETER

TYPE 001, 0.5 SEC. TIME CONSTANT

DRAWING NUMBER

process assures the precise physical matching of cavities by the use of a common mandrel on which to form the components and by using the integral of time and current to provide an exact knowledge of the mass of silver deposited on the mandrel. A discussion of the electroforming process will be given later in this report.

Other fabrication techniques also help maintain the necessary thermal symmetry. One of these consists of the careful use of silver and tin solder for joining components. In this process care is taken to use consistent solder weights and to minimize tinning of the silver on areas other than in the joints. Care is taken to "wet" these joints thoroughly and to eliminate joint gap spaces in order to provide the greatest possible heat transfer across joints.

Another design feature which aids in assuring thermal symmetry is the use of identical "postage stamp" type printed circuit thermopiles. An identical thermopile strip is wrapped around the thermal resistor of each opposing sensor to measure the temperature drop between the isothermal ring formed at the junction between the receptor cone and the thermal resistor cylinder and the heat sink temperature. Since all thermopiles are physically identical their effect on thermal resistor values are equal and symmetry is maintained.

IV Thermopile Design

The objectives of thermopile design were to obtain the greatest output voltage with minimum effect on time constant. Since the radiometer design incorporated a geometry symmetrical about the mounting disc it was also decided to use an identical thermopile on each side and to connect the thermopiles electrically in series opposition. This design differs from the present design which places the measuring junctions about the isothermal ring of the measuring cavity and the reference junctions about the isothermal ring of the compensating cavity. The new design doubles the actual number of junctions used but permits a wrap around application of a thermopiles "postage stamp" strip rather than having to route the between junction wiring through holes in the reference temperature mounting disc. This type of mounting of the thermopiles greatly simplified fabrication of the cavity assembly as well as enhancing the performance.

Since a printed circuit type of thermopile design was to be used, it was necessary to select junction materials having both high output and compatibility with the production methods used for printed circuit design. The first choice of materials from both high output and printed circuit compatibility was antimony and bismuth. The use of these materials would give double the output of a similar thermopile using the more standard chromel-constantan junctions. However, since there were problems associated with evaporating processing and toxicity of vapors from antimony and bismuth it was decided to go with chromel and constant for the present phase and to consider these materials in a later phase if sensitivity were found to be a major problem.

A local strain gage-thermocouple manufacturer, Micro Engineering, Inc. was selected to help develop the printed circuit thermopile. Since this development was considered to be the most difficult task of the project and since we had knowledge of the capability of this company we decided to work with them on this development. We submitted our desired specifications viz, maximum output, maximum number of junctions, thinnest backing material and minimal electrical resistance. We also provided drawings showing the desired geometry, connection means and possible junction form.

Micro Engineering came up with an alternative geometry and junction fabrication method which exceeded our expectations. The design utilized a wrap around strip which in the case of our short time constant sensor design would provide 176 pair of junctions per strip. This design, provided by Micro Engineering is considered proprietary by them.

Attempts were made to fabricate the thermopiles on 0.3 mil Kapton as a minimum thickness backing material. Both using multiple thermopiles per backing plate as well as using only a single thermopile proved too difficult to fabricate. A change to 1.0 mil backing material provided the manufacturing method to be possible but will result in poorer transient heat transfer to the junctions. A compromise using 0.5 mil Kapton backing appears feasible for the design presently being used. The electrical resistance of a thermopile of 176 junction pair is presently in the order of four thousand ohms which is still compatible with presently available amplifier input impedance requirements.

Three severe problems were encountered: weak bonding of the thermocouple material to the Kapton, the second thermocouple material would not release from the Teflon backing when transferred to the Kapton to form the thermopile junction, and the registration of the two parts of the thermocouple pattern. Each of these problems have been solved in other applications and are solvable for fabrication of the desired thermopile with additional time. However, because of the time constraint of this contract effort, work was stopped when sufficient information became available to determine feasibility and performance estimates.

V. Manufacturing Processes

A. Electroforming

Electroforming of metal is similar to electroplating with the exception that the plating is applied over a subsequently removable form or mandrel. By utilizing a mandrel of stainless steel and plating with a known current for a definite time, thickness tolerance may be held to the order of a few hundred thousandths of an inch. The removed work piece is known to retain very closely the shape of the mandrel on which it was formed thus making parts duplication accurate where symmetry or precision assembly is required. With subsequent annealing the desirable characteristics of silver such as high diffusivity and low thermal resistance are easily achieved. The elimination of hand fabrication, handling and reduced soldering requirements allow fabrication with significant reduction in thickness with a consequent reduction in mass.

The electroforming process was used to fabricate all of the cavity parts and shields. This minimizes the number of component designs required and assures a near balance of the sensing and compensating cavities to thermal transients. Only one solder joint is required to form each cavity and only one additional solder joint for attaching each cavity to the common mounting ring.

The principle material used in electroformed cavity parts is fine (99.99%) silver. When the silver is correctly deposited and annealed, it exhibits essentially the same characteristics as those of annealed rolled sheet silver. The fabrication process described assures uniformity of wall thickness, dimensional control and metal characteristics. It uses both proprietary

plating solutions as well as formulated solutions. At present the product quality is highly dependent on the use of proprietary solutions as well as on the use of pure silver anodes and the various plating and handling techniques described.

Mandrels of 17-4 PH or 304 stainless steel for the two parts of the cavity and the two shields were turned to dimension and given a high surface polish on those surfaces to be plated. The plating was stopped off to close dimensions at the cavity edges by means of tightly applied insulating bushings (washers) of Delrin. These insulators were removed after electroforming and prior to removal of the work from the mandrel. The mandrels are adequately tapered (approx. one to two degrees) for draft to assure easy work removal. The mandrel for the cavity shield contains a number of holes which were temporarily filled with insulating buttons to prevent the electroform from occurring where holes are desired. Holes appear in the work at these places since the plating will not readily form over insulating material. A short length of 0.500 inch diameter shank was left on one end of each mandrel for chucking, #10-32 tapped holes were added in blunt ends for both holding insulating washers on during plating and for handling rod attachment for use when high temperature furnace insertion was required.

Electroforming Process

During electroforming the mandrel to be plated was handled easily by the attachment of a short (2 inch) length of 10-32 threaded brass rod to the shank end. The rod tightly clamped the Delrin bushing to the mandrel for stopping off the plating. In the process, the unused end of this rod was chucked in a rotating chuck and used as a hangar above the silver forming bath. The threaded rod was waxed to prevent its being plated during the process. When a mandrel was thusly assembled it was ready for cleaning and plating. After cleaning the mandrel it was only handled by the support rod and was moved rapidly between solutions or suspended under clean water to prevent chemical changes to the mandrel surface. After an initial wash with liquid "Ivory" or equivalent and a thorough rinse under the tap in hot water, the mandrel was read to electroform. It was not allowed to dry and it was quickly transfered to the various cleaning and plating solutions of the process.

Process Solutions

For the electroforming process the following solutions were set up in clean glass beakers in a row starting from left to right on a suitable bench.

The proprietary solutions may be purchased from the EFS Company, Division of Tawa Traders, 3244 E. District St., Tucson, Arizona, 85714, telephone (602)573-1218.

Beaker No.	Solution	Temp.	PROCESS STEPS
			Remarks
1	#110 clean	125°F	Proprietary (or equivalent phosphate type solution, see alternate methods, Ref. 1)
2	Tap water	Room	
3.	KCrO ₃ + KOH	Room	Equal amounts of 1 oz of CrO ₃ (chromic acid) to 1 oz gal distilled water, and 1 oz of KOH (potassium hydroxide) to 1 gal distilled water.
4	#350 Activator	100°F	Proprietary - Approx. 300 cc in 500 cc beaker.
5	Dist. Water	Room	
6	Silver strike	Room	0.5 oz of AgCN, silver cyanide, per gallon of distilled water plus KCN, potassium cyanide, at 12 oz/gal water concentration. Use at 70 to 85°F 300. cc in 500 cc beaker. Use pure silver anode (No bag necessary). 15 to 15 amp/ft ² (~ 170 ma/inch ²)
7	Dist. water	Room	
8	Silver electroform	Room	Proprietary solution. Use at room temp. Use approx. 800 cc of solution in a 1 liter beaker in a pure silver anode cylinder around the inside wall of the beaker to assure even thickness deposition of silver. suspend the mandrel on the central axis. of the beaker.
10	Tap water	Room	

Ref. 1. ASTM Designation B 431-85, Standard practice for "Processing of Mandrels for Electroforming", PP. 1108-1115

Processing Times

The following processing times have been either experimentally determined or calculated. Silver in the electroform solution will deposit at the rate of one mil (0.001") per square inch per hour at a current of 43 milliamperes. From this, the deposition time to deposit a 5 mil (0.005") layer on each of the mandrels may be found. The following list shows the processing times for each beaker step.

Processing Time

Step	Solution	Time
1	#110 clean	3 minutes
2	Tap water	30 seconds
3	KCr ₃ + KOH	10 seconds
4	Distilled water	15 seconds
5	Silver strike	60 seconds
6	Distilled water	15 seconds
7	Silver electroform	By timer, see times below
8	Tap water	15 seconds

Deposition Rate of Silver

Silver deposits at the rate of 4.03 gm per ampere hour. At this rate the approximate ampere hours for deposition of the mandrels #1 through #6 was determined. The time required was then be calculated as determined by the current used.

Mandrel #	Area in Sq. "	Area Sq. cm	Grams of Silver	Ampers Hours	Deposit time at time at 0.1
1	1.25	8.06	1.08	0.27	2h 8m
2	1.62	10.44	1.39	0.35	2h 8m
3	4.34	27.98	7.47	1.85	4h 16m
4	4.64	29.90	7.98	1.98	4h 16m
5	0.84	5.43	0.72	0.18	2h 8m
6	0.94	6.07	1.61	0.40	4h 16m

Using 0.1 amp/square inch:

$$\text{Deposit Time} = \frac{\text{Ampere Hours}}{1/10 \text{ area in sq "}}$$

For other currents as may be required:

$$\text{Deposit Time} = \frac{\text{Ampere Hours}}{\text{Amperes} \times \text{Area in sq "}}$$

Removal of Workpiece from Mandrel

The use of a passivated stainless steel mandrel together with the use of an approximate two angular degree draft on long parts of the mandrel permitted easy removal of the workpiece. In order not to damage the very thin workpiece by physical removal from the mandrel, was used to aid in the process. This was done as follows:

After first removing the hangar rod and the stop off insulating washers, a long (possibly 8 to 10 inch) rod of stainless steel with thermally insulating handle was screwed into the tapped hole used for the hangar. Holding the mandrel by this handle (with glove protected hand) the mandrel was inserted into a furnace which had been preheated to approximately 1600°F. After a few seconds the silver electroform was heated from radiation and expanded. A slight "pop" was heard as the inner surface of silver part parts from the stainless steel mandrel. This differential expansion takes place since the poor heat conductivity of the stainless causes the silver to expand more rapidly relative to the stainless. After the electroform was loose on the mandrel it was subsequently annealed under controlled conditions.. The removed electroform part was handled and packaged with care to prevent damage

B. Printed/Deposited Thermopiles

The second manufacturing technique which we have explored is the use of thin film (printed circuit) technology for increasing the number of thermocouple pairs in the sensor thermopile. At present, the use of fine wire, chromel-constantan couples (which are spot welded and applied over the thermal resistor of the cavity sensor) limit the sensitivity which may be achieved using a reasonable number of junctions. Forty eight junction pairs compose the standard cavity assembly. With photo lithography techniques and thin film etching, the number of junctions in a given space will be significantly increased thus increasing sensor sensitivity. A factor of 3 to 4 increase in number of junctions has been shown to be feasible. The company now working with us on this problem and who we expect to produce an exceptional printed

circuit thermopile is Micro Engineering II, 14 N. Benson Ave., Upland, California, 91786, telephone (714) 946-2110. ATTN: John Hall.

VI. Test Item Manufacture

From subcontract item suppliers electroformed parts and printed/deposited thermopiles for two types of test cavity assemblies were ordered. In both cases, TMI personnel had to interact with the supplier in order to get parts which meet our requirements. Additionally, the cavity mounting disks were machined, and miscellaneous radiometer body and test apparatus were modified for use in the test radiometers and testing.

The cavities assemblies were fabricated through a process previously described. The thermopiles were installed on the cavity assemblies by techniques standard in strain gage installations. That is, 610 cement was used with the thermopile held in place under pressure and the adhesive cured in an oven at elevated temperature. Lead wires were attached to solder pads on the thermopile sheet. The cavity was then installed in a test radiometer body. Sufficient test radiometers and apparatus were fabricated to support the test program required to verify the sensitivity and time constant predictions.

VII. Test Program

The test program consisted of two parts for each cavity type. First, response time was measured. On the test radiometer was installed a high speed shutter. The radiometer was then mounted on a tracker indexed on the sun. A "fast response" XY recorder was hooked up to the thermopile leads. After adjustment and stabilization, and with a "steady sun" the shutter was tripped and the response recorded. This type test was performed a number of times for each cavity type. When repeatability of response time measurement was demonstrated, the test series was terminated and the next test performed.

The sensitivity measurement test required more rigorous test control. For these tests, a measurement of the incoming solar radiance is required. The TMI Standard Instrument, Mk VI S/N 67401, was also installed on the tracker.

Simultaneous measurements of incoming irradiance and radiometer response recorded. These test were repeated until repeatability was demonstrated.

The test equipment used are as follows:

1. Gould Series 60000 XY recorder, S/N 052134

Main Frame

Writing speed: 120cm/sec in Y axis, 60cm/sec in X axis

Slew speed: 134cm/sec

Linearity: $\pm 0.1\%$ f.s.d.

Repeatability: $\pm 0.1\%$ f.s.d.

X & Y Input Amplifiers

Input impedance: 1Meg ohm

Sensitivity: 18 ranges from 0.05mV/cm to 20.0V/cm

Time Base

Speeds: 8 ranges from 0.1 to 20.0sec/cm

Accuracy: $\pm 1\%$ f.s.d.

Linearity: $\pm 1\%$ f.s.d.

2. Data Precision Digital Multimeter, Model 3600, S/N 1678

Specifications for .2V range

Resolution: 1uV

Input Impedance: >1000Meg ohm

Accuracy: 24Hr: $\pm 23^{\circ}\text{C} + 1^{\circ}\text{C}, \pm (0.005\% \text{ inp.} + 4 \text{ l.s.d.})$

Common Mode Rejection Ratio: 160 dB at dc with 1000 ohm
unbalance

3. Technical Measurements Radiometer Control Unit, Mk I, S/N 47302

Specifications when used with Mk VI Radiometer

Accuracy: Absolute measurement uncertainty is less than
 $\pm 0.5\%$ of the TMI radiometer range

Zero Drift: $< \pm 0.1\%$ / month of TMI radiometer full range

Time Constant: < 2 seconds (1/e)

Readings in Absolute Energy Units

RESULTS OBTAINED

I. Analysis and Manufacturing Results

Our analysis indicates two principal things. First, the required time constant for the cavity sensor is probably longer than the requirement statement of two seconds, and second, a cavity sensor with a time constant of less than two seconds can be designed that will have the required sensitivity. The parameter chart given in Figure 3, page 16, shows that one can trade time constant for sensitivity. Because a cavity can be designed with a time constant of significantly less than two seconds, consideration can be given to adding an electric heater to the cavity cone for self calibration. The added mass of the heater can be tolerated with a corresponding increase in response time without undue reduction in sensitivity.

In the manufacturing phase of the work, good success was obtained in the electro-forming of the cavity parts in silver. The density of the deposited silver was close to that of sheet stock and good control of thickness was achieved. No problems in fillet buildup or thin spots occurred when proper procedures were followed. Clearly, electroforming of the cavity sensor parts yield the satisfactory silver parts for the cavity sensor. The assembly process using oven soldering techniques worked well.

Sufficient silver cavity parts and copper mounting rings were fabricated for assembly of test cavities of two types: the standard Kendall design (designated PR for PACRAD radiometer) modified for the new manufacturing method, and the new design for high sensitivity and fast response (designated HSFR).

The fabrication of the thermopiles was an entirely different matter. We first had to back away from antimony-bismuth thermocouples because no R&D fabricators were available who could handle the health/safety hazards associated with these materials. It then became apparent that while all the fabrication steps needed to produce a successful thermopile had been developed and used in the fabrication of other products, the combination of them in producing a single product had not been done. The dimensional stability of the Kapton during the fabrication processes caused problems in the adhesion of the thermocouple material and registration of the patterns of the two materials. Improved control of ambient temperature and holding tension of the

Kapton should cure this problem. To make the thermocouple junction, it is necessary to vacuum deposit a second material on to a plate which is then thin film etched to develop the desired wire pattern. The resulting wire pattern is then transferred with proper registration to the Kapton sheet containing the first material. In this process, difficulty was experienced in the release of the thermocouple material from the Teflon backing material. The cause of this is not known at this time, but it is probably caused by surface activation of the Teflon during the vacuum deposition process. This technique has been successful in other applications.

In the many attempts made to fabricate the thermopiles, partial successes were achieved but no functioning thermopiles were obtained. Additional development work is required which will take many months. To continue the development work on the cavities, and allow estimates of performance to be made, we resorted to building thermocouples and thermopiles out of small (36 gauge. 0.005" diameter) thermocouple wire.

II. Test Results

A. Sensitivity

The test results of the measured sensitivity in ambient air are quite good. The cavities were instrumented with 14 junction thermopiles made out of wire and mounted on Kapton insulation. The measured results were then ratioed up by a factor of 174/14 to account for the fewer junctions of the wire thermopile. The Kendall standard design (PR) indicated a sensitivity of 32 $\text{uv}/(\text{mw}/\text{cm}^2)$ and the high sensitivity fast response design indicated 60 $\text{uv}/(\text{mw}/\text{cm}^2)$. In vacuum, the sensitivity should increase approximately 30%. The sensitivity of the HSFR cavity exceeds the requirement statement of 0.02' $(\text{mw}/\text{cm}^2)/\text{uv}$. The PR cavity would just meet the requirement in vacuum (space).

B. Response Time

The nature of the 14 junction thermopile used for making the sensitivity measurements, preclude making short response time measurements. Because of the thermal insulation of the Kapton and the relative massiveness of the

wires, the measured time constant is dominated by the response time of the thermopile. This is evident in the measurements of response time for the two cavity types when instrumented with the 14 junction thermopiles. The PR cavity exhibited a response time 2.3 seconds, while the HSFR cavity showed a response time of 3.4 seconds. The installation of the thermopile is more difficult on the HSFR cavity and resulted poorer thermal conductivity at the junctions to the silver thermal resistor. To overcome this problem, a single thermocouple junction was attached by solder to the thermal resistor with as small a bead as possible. For good attachment, the resulting bead size was 0.02". With this thermocouple, the response time of the HSFR cavity was measured to be 1.85 seconds. This meets the requirement statement of less than two seconds.

III. Performance Estimates

The test results obtained in the research, to this time, are compromised by the lack of an adequate thermopile. The bead size used with the HSFR cavity determined the measured time constant. The Omega Temperature measurement Handbook and Encyclopedia provides data on response time verses bead size. This data suggests a thermocouple junction of the size we used to about 1.4 seconds. It also suggests that a reduction in size to our target size of 0.003" would result in a junction time constant of less than 0.1 seconds. The overall time constant of the cavity sensor would then be very close to the time constant of the silver parts. Based on these considerations, we estimate that a HSFR cavity can be produce with a sensitivity of 0.02 (mw/cm²/uv) with a time constant of less than two seconds.

ESTIMATE OF TECHNICAL FEASIBILITY

In the short time TMI had to conduct this research, it was necessary to make estimates of the desired characteristics of the sensor and then to build models which thorough testing would demonstrate the design principles and allow predictions of performance. From that base, the design of optimum sensors for the specific ACAI mission can be developed. The results of the Phase I work shows that cavities tailored to the specific requirements of each ring of the ACAI can be provided.

CONTRACT REPORTING REQUIREMENTS

I. Patent Rights & New Technology

No inventions or new technology were developed in the course of this work. Proprietary processes and techniques were utilized by subcontractor item suppliers of fabricated parts.

STATEMENT CERTIFYING LEVEL OF EFFORT

The amount of effort expended in performing this contract is as follows:

LABOR CATEGORY	DIRECT HOURS EXPENDED	DIRECT HOURS ESTIMATED
Officer in Charge	51	84
Professional Staff	423.5	392
Support Staff	41.5	156
Clerical/Typing	51	<u>64</u>
TOTAL	567.5	696

Based on the burdened hourly rates per labor category as show in H.5 of the contract, the total fixed price of this contact should be reduced by \$5,082.50 to a value of \$44,917.50.